

Division FALCON

DIRECTION GÉNÉRALE TECHNIQUE

Mérignac, October 27th, 2004

FAA Transport Airplane Directorate, ANM-116 1601 Lind Avenue,

SW, Renton

Washington 98055-4056

USA

Attention to

Rules Docket

(Fax (202) 493-2251)

No 2003-NM-30-AD

DGT-DTF/NAV 298060 CHG/PP

Please contact

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SUBJECT: NPRM Docket No 2002-NM-202-AD - F10- second Comments:

Cracks in nose landing gear strut actuator

References

[Doc1] MCAI form DGT-DTF/NAV 85738 Dated May 24,2002, sent by note DGT-

DTF/NAV 272240 dated. May 24, 2000 ' Nose gear slow retraction.'

[Doc 2] DGT-DTF/NAV 297656 same title

Dear Sir,

This note is a complement to [Doc 2] on an other paragraph.

The paragraph titled "Part Installation", paragraph [m] reads:

"As of the effective date of this AD, no person may install on any airplane a drag strut actuator having part number 747721."

If the NDT, ultrasonic inspection for cracks in the actuator piston rod is done without any evidence of a crack, it is acceptable to allow the actuator to remain in service on the aircraft. A person reading this paragraph could conclude that it is unacceptable to install a P/N 747721 actuator, if the actuator is removed from the aircraft for ANY reason.

The recommended change (addition), to paragraph [m] is:

" As of the effective date of this AD, no person may install on any airplane a drag strut actuator having part number 747721, which has failed the ultrasonic inspection of the piston rod for cracks, in accordance with Dassault Service Bulletin F10-294, dated March 20, 2002."

Do not hesitate to contact us for further information.

Best Regards,

Att: as described in the text

C. GIRAUDEAU



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EXTERNAL COPIE(S):

Mr RODRIGUEZ FAA ANM-116 Renton Mlle S. MORALES SFACT/N.AT Paris M GATARD GSAC Mérignac